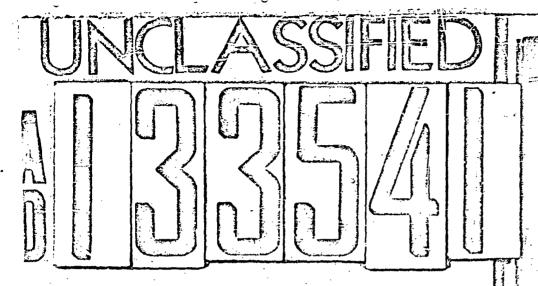
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U. S. NAVAL RADICIOGICAL DEFENSE LABORATORY
SAN FRANCISCO 2L. CALIFORNIA

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JUN 2 1 1957

From: Commanding Officer and Director

To: Department of Defense Agencies on Distribution List for Report

Subj: U. S. Naval Radiological Defense Laboratory Report USARDL-TR-139; forwarding of

Encl: (1) U. S. Naval Radiological Defense Laboratory Report USARDL-TR-139 entitled "A Fallout Forecasting Technique with Results Obtained at the Eniwetok Proving Ground"

- 1. A method for forecasting fallout areas was developed to supply the operational needs of scientific projects at the Enimetok Proving Grounds. The method employs a simplified cloud model and distribution of particles in the cloud based on parameters obtained from previous meapons tests and from theory. The particles are tracked to the earth's surface by considering their falling speeds and effects of the winds existing aloft. Tracking is simplified by use of the plotting device described in a previous report.
- 2. The end product of this technique is a plot showing the perimeter of the fallout area, the "hot line", or line through the center of areas of highest radiation intensity, and the time of arrival of fallout throughout the area. The method does not yield the intensity of radiation in the area. However, relative intensities can be surmised assuming a radiation gradient from the hot line to the perimeter.
- 3. Fallout forecasts made with this technique for four r ats at a recent operation are compared with areas of measured fallout. The comparison is excellent and it is evident that the technique works as well for mater surface as for land surface detonations.
- 4. This method is simple, first, and suitable for field use. It can, therefore, be adapted to the needs of military forces in the field.

BICHARD S. MANDELKORN

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A FALLOUT FORECASTING TECHNIQUE WITH RESULTS
OBTAINED AT THE ENIWETOK PROVING GROUND

Research and Development Technical Report USNRDL-TR-139

NS 081-001

3 April 1957

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ABSTRACT

A generalized fallout forecasting technique is presented with detailed computations of input parameters which were used at the Eniwetok Proving Cround.

Results obtained at a recent weapons test are briefly discussed by comparison of forecast fallout with preliminary measured data.

SUMMARY

The Problem

A fallout forecasting technique is needed to qualitatively describe the fallout hazard resulting from nuclear detonations. This technique should have such flexibility that its employment is valid for field use.

Findings

A summary of the latest experimental and theoretical considerations has resulted in the development of a technique whose complexity is dependent on the required accuracy of the results desired. Such a technique has been satisfactorily tested at the Eniwetok Proving Grounds for land surface and water surface bursts.

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ADMINISTRATIVE INFORMATION

This work was done under Bureau of Ships Project No. NS 081-001, Subtask 1, Technical Objective AW-7, as described in U.S. Naval Radiological Defense Laboratory Annual Progress Report (DD Form 613) to the Bureau of Ships, July 1956.

The fallout studies were made at Operation REDWING, Project 2.6.3, as described in DD Form 613, NS 088-001, Subtask 4B, Encl (1) to CO USNRDL Secr ltr 3-905-335 Ser 0014173 of 16 March 1956.

The work also is part of the technical program for the Department of the Army established between Department of the Army, Office, Chief of Research and Development, and Bureau of Ships (Joint Agreement, 23 November 1955).

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I INTRODUCTION

Fallout research continues to seek a theoretical working model that will describe in detail the mechanism of fallout. Aside from this long-range problem, consideration must be given to making available a working tool that will meet the needs of the military for solving fallout problems in the field. Such consideration requires a simplified rapid system capable of producing qualitative if not quantitative results.

Within a program studying fallout at a recent weapons test operation there was a fallout forecasting assignment that had many aspects of the practical field problem yet, at the same time, required quantitative results for use in reducing other data. This program needed positioning data such that three ships could be located properly in the fallout to obtain data on its parameters. Also, aerial and oceanographic survey projects required knowledge of the fallout to instigate their navigational procedures properly.

To meet these requirements a technique for rapid fallout forecasting was developed which not only satisfied the needs of the fallout program but also was accurate enough to allow comparison between the meteorological aspects of model work and the results obtained from surface measurements. This technique was restricted to describing quantitatively the perimeter of the fallout, the axis of the "hot line," and to determining the time of arrival of fallout throughout the pattern. No attempt was made to quantitate the expected levels of gamma activity or to develop radiation contour lines.

At this operation the Task Force employed a fallout prediction unit for determining the safe time to detonate the test devices. Although many of their techniques for forecasting were similar to those described in this report, their problem was of a different nature than that of the fallout program. Several of their methods were unique in that portable analog computers were tested as field instruments. These computers permitted consideration of many complex parameters. One, in particular, obtained essentially an instantaneous solution to the problem once the meteorological data were available.

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The fallout program and the lask Force prediction unit functioned independently. It was not feasible for the two to employ the same technique because the post shot variability of the winds aloft were especially critical in ship location problems of the fallout program. This problem will be discussed in detail later.

1.1 Objective

This report describes a technique for forecasting fallout employed at a recent weapons test operation. The results obtained in the field are discussed as examples of the reliability of the techniques. Although the technique was designed for analysis of land surface detonations where the fallout is particulate, its application to water surface detonations is considered.

2 FORECASTING TECHNIQUE

The forecasting technique uses many ideas from fallout model work. Several simplifications, as well as a plotting device, have been developed to the end that the time involved has been reduced greatly without sacrificing accuracy. In general, an initial source of activity is defined, describing the "stabilized" nuclear cloud by appropriate spatial and size distributions of radioactive particles. These particles are tracked to the earth's surface by considering their falling speeds and effects of the winds existing aloft.

2.1 Basic Considerations

In some cases the input parameters for the forecasting technique were obtained from weapons test measurements. In others, where data were lacking, the parameters were derived from theory.

2.1.1 Source Model

The optical or visible dimensions of the initial cloud from a nuclear detonation have been documented in past weapons tests. Available data describe such parameters as height to base of mushroom, height to top of mushroom, and mushroom diameter as functions of time. Vertical rise stabilizes in approximately 6 min post detonation. This time is independent of yield however, the expansion of the mushroom diameter, particularly for the megaton devices, continues for perhaps 30 min. Available diameter measurements have not been made in excess

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of H+10 min, however fairly reliable data are known for the optical cloud dimensions as functions of yield to H+10 min. The ultimate cloud diameter can be exirapolated from low yield curves and some qualitative data. Figures 1 and 2 present values of the cloud dimensions from past tests. The source model was assumed cylindrical having, for a given yield, these dimensions. Its stem diameter was taken as 10 percent of mushroom diameter.

2.1.2 Activity Distribution in Source Model

The greater part of the artivity was assumed to be concentrated in the lower third of the musinroom. The lower two-thirds of the stem was ignored; the remainder of the stem and upper two-thirds of the cloud were weighted lightly. This description (Fig. 3) of the activity distribution within the cloud appeared most reasonable in the light of available data and logical theoretical considerations. The activity was concentrated nearer the axis of symmetry of the cloud than at its outer edges.

2.1.3 Particle Size Distribution in Source Model

All particle sizes were assumed at all elevations within the cloud except the lower two-thirds of the stem. However, to obtain agreement with past fallout measurements and with the optical diameter of the mushroom, it was necessary to fractionate the particle size distribution radially within the cloud. Otherwise the computed fallout area about ground zero would be too large. The fractionation was specified as follows: particles of 1000 microns in diameter and larger were restricted to the inner 10 percent of the mushroom radius or approximately the stem radius; those from 500 to 1000 microns in diameter were limited to the inner 50 percent of the cloud radius. Since the relation of activity to particle size is some direct function of the particle diameter, this fractionation tends to concentrate the activity about the axis of symmetry of the cloud.

2.1.4 Particle Falling Speeds or Settling Rates

Computations of the terminal velocities of the particles were based on aerodynamic considerations for a still atmosphere having temperature and density distributions typical of the Marshall Islands atmosphere in the spring months.

Experimental data from past tests at Enswetck Atoll indicated that the particles were irregular in shape and had a mean density of 2.36 g/cq cm.

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[·] Darwhished data from a recent weapons test.

It can be shown that particles falling at their terminal speed experience three types of flow in a fluid; streamline or laminar flow where viscous forces predominate $(10^{-6} \le R_e \le 2.0)_T$ intermediate flow where inertia forces predominate $(2 \le R_e \le 500)_T$ and turbulent flow where inertia forces predominate $(500 \le R_e \le 10^3)_T$. Below a Reymolds number of 10^{-6} certain corrections must be applied to the equations because the particle diameter approaches the mean free path of the fluid medium; the region above a Reymolds number of 10^{-6} is important only in ballistics. These limiting cases will not be discussed here.

The parameters actively affecting a particle's falling speed are; its weight; its drag coefficient; its density; as well as the fimd density and fluid viscosity.

Most empirical equations developed in past experimental work have been for spheres dropped in various liquids. Some work has been done on irregular-shaped particles and some done in wind tunnels. The equations used to determine the falling rates for particles in a find medium follow.

For streamline motion, 16"4 4 Re 42.0

$$V_{s} = K_{s} \left(\frac{\rho}{\rho} \right)^{-1} \left(\frac{\mu}{\rho} \right)^{-1}$$
 (1)

where

Vs = terminal velocity in cm/sec

= particle density in gm/cm3

= fluid density in gm/cm3

d = particle diameter in cm

u = absolute viscosity of fluid in poises

Ks = constant incorporating gravity

= 54.5 for spheres

= 36.0 for irregular-shaped particles.

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The limiting diameter to which Eq (1) holds is:

$$d' = \left(\frac{36\mu^{l}}{g\rho_{0}(r^{2}-\rho_{0})}\right)^{1/3}$$
 for spheres and
$$d' = \left(\frac{54.4\mu^{l}}{g\rho_{0}(\rho_{0}-\rho_{0})}\right)^{1/3}$$
 for irregular-skaped particles.

For Intermediate motion, 2.6 4 Re 4 500

$$V_{1} = K_{1} \left(\frac{2 \rho_{0}}{\rho_{0}} \right)^{2 h} \left(\frac{\mu}{\rho_{0}} \right)^{2 h} d_{0}$$
(2)

where

do = d - 7d'

= 0.4 for spheres

= 0.279 for irregular shapes

i' = limiting diameter to which streamline motion applies

K₁ = 30.0 for spheres

= 19.0 for irregular-shaped particles.

The limiting diameter to which the Eq (2) beins is:

$$d'' = 43.5 \left(\frac{\mu^2}{g \rho_0} \left(\rho_{-\rho_0}\right)^{\frac{1}{3}}\right) \text{ for spheres}$$

$$d'' = 51 \left(\frac{\mu^2}{g \rho_0} \left(\rho_{-\rho_0}\right)^{\frac{1}{3}}\right) \text{ for irregular-shaped particles.}$$

For turbulent motion, 504 &R. \$105

$$V_{T} = K_{T} \left[\left(\begin{array}{c} P_{-} \\ P_{0} \end{array} \right) c^{\frac{1}{2}} \right]^{\frac{1}{2}}$$

$$K_{T} = 54.6 \text{ for spheres}$$

$$= 50.0 \text{ for irregular-shaped particles.}$$
(3)

This equation taken from Ref 1. However, certain constants have been re-evaluated.

These equations were taken from Ref 1, However, certain constant have more re-evaluated.

The average falling rate for a group of irregular-shaped particles of a given size will be given by the equations. However, individual particles of the group may deviate from this average.

2.1.5 Marshall Islands Atmosphere

Marshall Islands atmospheric conditions determined the values for the density and viscosity parameters used in computing particle falling rates. Available data on the temperature, pressure, density and viscosity as functions of altitude for the atmosphere common to the Marshall Island area in the spring months follow.

It was not possible to use a "standard atmosphere" in this problem because such use introduced a large error in the particle falling rate at high altitudes. This error originates primarily because an isothermal layer is assumed above the tropopause in the standard atmosphere—an unrealistic assumption.

Temperature Distribution. From the weather data published by Task Force Weather Certral at Operation CASTLE, four published radiosonde runs obtained temperature measurements to high altitides:

1 March 1954 0600 M Bikini

27 March 1954 0600 M Bikini

7 April 1954 0620 M Bikini

26 April 1954 0610 M Bikini

No data were available above 67,000 ft. Fortunately two of these runs pen-trated the tropopause which was located at approximately 55,000 ft. To extend the measured data beyond 67,000 ft, climatological averages? for latitude 12°N were employed. Agreement with measured data was satisfactory except for the range from 50,000 to 55,000 ft where the climatological data indicated a well-defined isothermal layer. The most significant finding from the measured data was the complete lack of an isothermal layer above the tropopause. Instead, a distinct and rapid inversion was observed which, when extrapolated as a straight line, agreed with the climatological data above 70,000 ft. Since the atmosphere was to be defined to 120,000 ft, further extrapolation was necessary. Temperature data available at these higher altitudes were taken by rockets' over White Sands, New Mexico. A plot of three points from the rocket data justified to some extent a continued extrapolation of the curve to 120,000 ft.

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Therefore the profile of the vertical temperature gradient (Fig. 4) was based on measured data to 67,000 ft and extrapolated to 120,000 ft on the basis of supporting climatological data and temperature measurements made at high altitudes with rockets.

Pressure Distribution.

of the pressure distribution were obtained on two occasions at Operation CASTILE. These measurements," made at Bikini on 7 April 1954 and on 26 April 1954, were not taken above 65,000 ft. Above this altitude the pressure was extrapolated as a straight line on semi-log paper to 120,000 ft. Agreement with published rocket data from White Sands, New Mexico was good to 92,000 ft (Fig. 5).

<u>Density Distribution</u>. The density distribution of the atmosphere (Fig. 6) was calculated from the perfect gas law using the above pressure and temperature distributions.

$$\rho = \frac{P}{RT}$$

where the gas constant was taken for dry air. The assumption of no moisture in the mixture introduces an error of several percent in the lower layers of the atmosphere where the relative humility is high. However, this assumption can be safely neglected. Also, it has latest theories on the composition of the atmosphere indicate it to be constant to altitudes above 150,000 ft which justified the assumption of a non-varying gas constant.

<u>Viscosity Distribution</u>. The variation of absolute viscosity with altitude was computed from the observed temperature distribution using Sutherland's formula,

$$\mu = \mu_0 \left(\frac{T_0 + 114}{T + 114} \right) \left(\frac{T}{T_0} \right)^{3/2}$$

$$\mu = 0.01769 \quad \left(\frac{387.17}{t_1 + 114}\right) \left(\frac{t_1}{273.17}\right)^{3/2}$$

where t_i = temperature in degrees Kelvin and μ is viscosity in centipoises. These data are plotted in Fig. 7.

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^{.*} Hq. T. U. -13 Operation Metro No. 14, 32 April 1958.

The data on pressure, temperature, density, and viscosity in 1006-ft intervals to 120,000 ft are summarized in Table 1.°

2.1.6 Terminal Velocity Computations

The average falling speed through 5000-ft layers was computed for four particle sizes over an altitude range from 0 to 120,000 ft. In these computations all in-flight transition of the particles from streamline to intermediate flow had to be considered through use of the plot shown in Fig. 8.

Four particle sizes (75, 100, 200, and 350 μ diameter) were employed since there was evidence from past tests that the 75- μ particle defined the limiting distance of failout of interest and the larger sizes best described the pattern within this limit. Table 2 presents the falling speeds computed for the four sizes. Tables 3, 4, 5, and 6 display the cumulative time of fall from a given altitude for these particle diameters.

2.1.7 Meteorological Procedures

It is necessary to have available the best possible description of the winds aloft in order to determine the arrival points of particles of various sizes originating at various altitudes. Such data are usually availabel from the normal upper air soundings routinely taken by Weather Bureau and Military Meteorological stations. Although wind velocity as a function of height varies continuously it can be described by an average speed and direction in discrete layers. Such averaging can best be obtained from the WBAN-20 Form where the original data are recorded. The technique employed in this report was to divide the atmosphere into layers 5000 ft thick and determine an average speed and direction for each layer. When the average falling speed of particles through these 5000-ft layers and the speed and direction of the wind are known, horizontal displacement can be computed. Thus, for each particle size a vector may be drawn for the average particle displacement in a particular 5000-ft layer. Addition of such vectors from all layers described the trajectory projection of a particle of given size. Similar plotting for ail particle sizes originating at all elevations within the cloud source will map the fallout on the earth's surface.

This technique is valid for any atmosphere that has negligible vertical motion and is in a steady state condition with respect to the horizontal winds during the time needed for the slowest particle to fall from the highest altitude to the ground. Such an assumption is not realistic for situations arising from many of the megaton devices because 15 to 20 hr are necessary to establish the fallout area. Consequently, when computing particle trajectories, an attempt should be

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made to consider how the wind varies with time and how it varies with distance from ground zero; what effect vertical motions have on particle falling speeds and how they vary with space and time. Such considerations complicate computation of trajectories extremely. In most cases valid input data describing these variables are not available. This phase of the problem is discussed below.

2.2 Plotting Technique

The use of "particle-size" and "beight" lines in mapping fallout is a standard technique employed by most analytical methods. This technique simply describes a grid (Fig. 9) on the earth's surface indicating where fallout particles of certain sizes will arrive and from what altitude they came. These parameters are the basic data for describing the fallout pattern.

Assuming steady state meteorological conditions without vertical motion or space variation of the winds, it is very easy to construct a grid describing arrival points on the earth's surface for particles of various sizes originating at different altitudes. This grid is constructed by ignoring the horizontal distribution of particles in the cloud model and by plotting those trajectories that originate along the line source describing the vertical axis of the cloud.

Plotting trajectories for each particle size at every starting elevation is the first step in determining the resultant fallout pattern; however, the drafting involved is tedious and time-consuming. This effort can be reduced greatly by plotting from the ground up, as is done in the construction of a wind hodograph. Such a plot is made by starting at ground zero and working up through the altitude increments to the desired elevation. Although this technique does not plot the trajectory of the particle, it does define the arrival points on the surface of the earth of particles starting at each altitude increment (Fig. 10). To plot these size-lines one must make the preliminary computations of particle-falling times through each altitude increment to obtain the displacement for various wind velocities as described earlier in the Section on Terminal Velocity Computations (p 8).

A plotting device (Fig. 11), described elsewhere, facilitates the computations required for the size-lines of the fallout pattern. Such devices were constructed for four particle sizes: 75, 100, 200, and 350 µ in diameter. With these plotters, trajectories or size-lines can be plotted from any elevation up to 120,000 ft for the four particle sizes. The plotters automatically account for the variable particle falling speed. They also eliminate the need for drafting equipment. After establishing the particle arrival points by either the use of size-lines or trajectories, height lines can be constructed. These lines, joining

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A great deal of excellent upper air data for the Marthali limit was obtained at Operation REDWING
in 1956. Reduction of these data will result in a much better description of the Marthali Islands
atmosphere than has been previously available.

surface zero with the arrival points of all particles from the same elevation, are most descriptive for they define the gath along which all particle sizes will deposit from that originating altitude.

The height lines describing the fallout from the lower portion of the mushroom immediat ly establish the "hot line." The "hot line" is best defined as that portion of the fallout area wherein the highest levels of activity are found relative to the adjacent areas. Under most meteorological conditions this area is described by a line from surface zero that coincides with the height lines from the altitude layers that include the base of the mushroom; for the source model was so defined to concentrate the activity in this volume.

Since the plotted grid of size-lines and height-lines was based on a line source of activity, each particle point must be expanded to the appropriate cloud or stem diameter from which it originated. This expansion, after taking into consideration the radial particle size fractionation in the source model, defines the perimeter of the area. One then has a map indicating the fallout area and the path of expected highest activity.

Curves of time of arrival of fallout through the pattern are established by simply assigning the appropriate value of falling time to each expanded circle about the arrival points and by constructing from this network of values iso-time contours that indicate the earliest time at which fallout will arrive at a given distance from the shot point. Similarly, the determination of the time of cessation of fallout at any location may be plotted. However, one is faced with the question of how to define cessation. Very small particles that do not contribute significantly to the radiation field continue to arrive for days after time zero. Consequently, a plot which describes time-to-peak activity seems more meaningful. During the field operation time-to-peak activity was defined as the time of arrival of fallout particles originating in the lower third of the mushroom.

This method determines the fallout plot under conditions that do not involve several important meteorological variables. In this sense it is most valia for a fallout of short duration and over a relatively small area, for-example, a l-KT surface detonation. Megaton devices and large KT yields deposit primary fallout over long periods and to great distances. To map such extensive deposition of fallout necessitates inclusion of complex meteorological variables and consideration of the fact that clouds from these large detonations extend to great heights in the atmosphere.

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2.2.1 Time Variation of the Winds Aloft

In most of the observations made at the Entwetok Proving Ground the winds aloft were not in a steady state. Significant changes in the winds aloft were observed in as short a period as 3 hr. This variability was probably due to the fact that proper fixing conditions, which required winds that would deposit the fallost north of the proving ground, occurred only during an unstable synoptic situation of rather short duration. It was necessary to correct for this variation to keep track of the predicted fallout area, especially at great distances from surface zero where as much as 20 hr elapsed before deposition.

Since this variation could not be forecast, balloon runs were made every 3 hr from H40 to H+24 and each particle trajectory employed the winds as they changed with time. The correct particle trajectories were approached by a method of successive approximations as follows: Tables 3 through 6 were computed for the four particle sizes and gave their cumulative times of fall such that starting at any elevation their altitude at any time after H-hr could be located. For example, the 75-µ particle originating at 70,000 ft entered the 40,000-ft layer in 7,18 hr and reached the surface in 19 hr. Since new upper air observations were obtained every 3 hr it was assumed that the balloon released at H+0 represented the winds until H+6 hr and so on. Therefore, as the particle settled to earth the appropriate winds aloft were applied to it.

The first step was to plot size-lines for the particles based on the H+0-hr winds. This established a fallout plot that assumed the winds would not change with time. When the 14+3-hr winds became available a similar plot was made based on them. With the aid of Tables 3 through 6 the particles starting at various elevations were located in altitude at H+3-hr points. These H+3-hr points are marked at the proper altitude on each size line. The two size lines, Hr0 and H+3, are then overlayed such that the H+3-hr points are coincident and the combined size-lines determined with the aid of a light-table. This is done by taking the upper portion of the H+0-hr size-line and the lower portion of the H+3-hr sizeline. This first approximation then assumed that the H+3-hr winds will remain steady for the remainder of the particles flight. The process is repeated using the combined size-line and the new size-line for the next set of wind data until the particle reaches the surface. Therefore for each new wind observation a closer approximation of the corrected time variable plot is made until ultimately the plot is quantitative.

2.2.2 Space Variation

The preceding computations assumed that the winds aloft, as measured at the point of detonation, at a given time are the same throughout the area for that time. Since the fallout can deposit hundreds of miles

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A recent study of available Asta indicates that the time-to-peak activity can be excellently defined as twice the time of arrival.

from surface zero, the ideal situation would be to take winds-aloft measurements throughout the volume traversed by the particles. Correction for space variation of the winds is then necessary, however in most cases not as significant as is time variation. Most weather networks are not refined enough to allow quantitative correction for these errors.

2.2.3 Vertical Motions

In applying particle falling speeds to the forecasting technique, it is assumed that the atmosphere has no vertical velocity. Computations made at the Eniwetok Proving Ground* to 50,000 ft indicated that large cellular vertical motions in the atmosphere sometimes attained speed: equal to and greater than the settling speed of a 75-11 particle. A time-space correction should be madefor the falling speeds of the particles to compensate for this parameter. However, in the work at the test site it was not possible to include this effect in the fallout forecasts. Certain anomalies discussed below may be due to such an effect and post shot analysis is being conducted to see whether they are resolved when the vertical motions have been taken into account.

3 DISCUSSION OF FIELD TEST RESULTS

The forecasting technique described was employed by the fallout program at the Eniwetok Proving Ground to satisfy certain project requirements. One project had three ships equipped to collect fallout and their positions had to be determined for most efficient collection; another sampled the ocean for fallout; while another made an aerial survey of the contaminated area. The navigational schedules for these latter projects were based on the forecast fallout pattern. Operations were controlled through the Program Control Center aboard the Task Force Command Ship where the forecasts were prepared.

The meteorological data were received from the weather ship at Bikini atoll as well as from weather stations at Rongerik atoll and Eniwetok atoll. Furthermore, all forecasts made by the Task Force Weather Central at Eniwetok atoll were usually available aboard the command ship by facsimile through the ships Weather Station.

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Upper air measurements were made at Bikini, Rongerik and Eniwetok atolls every 3 hr starting at H-24 hr and continuing until H+24 hr for any given cetomation. The frequency of observations was usually increased during the period from H-6 to H-2 hr. The altitudes reached on the wind runs were remarkably high and gave perhaps the best set of winds-aloft measurements to date. The average termination altitude was approximately 50,000 % with many runs over 100,000 ft. Such excellent coverage of the winds aloft was a major belp in the fallout forecasting.

Fallout forecasts were made every 3 hr starting at H-24 hr, using the messured winds available at the time. This process was continued up to shot time and from then, on the technique of cornecting for time variation, was employed every 3 hr until the fallout event was completed. It was not feasible to correct for space variation and vertical motions during this period because of the lack of time and data.

3.1 Fallout Plots

The fallout forecasts determined at the weapons test operation were based entirely on measured data and quantitatively considered time variation of the wind. No space variation corrections or computed values of vertical motions were employed in their construction.

The area of measured fallout from shot A is compared with the forecast fallout plot in Fig. 12. Figures 13, 14, and 15 are similar comparisons for shots B, C, and D. Although C and D were water surface shots, it is evident that the forecasting technique succeeded in representing the measured fallout area as well as it did for the land surface detonations, A and B.

The comparison is excellent for all shots except B and as yet the discrepancy between the forecast fallout area and that which was measured in unknown. There is some indication that consideration of vertical motions will have to be made for shot B during the time of fallout since computed vertical motions were significant in magnitude. Such analysis including space variation is being carried out at this time for all four detonations and the refined data will be published later.

4 SUHMARY

The fallout forecasting technique described in this report was successfully employed for both land, surface and water surface detonations

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Under the direction of CDR Daniel F. Rex. Joint Task Force Seven, Meteorological Center, Fearl Harbor, T. H.

at the Eniwetok Proving Ground. With known meteorological data such a technique will successfully quantify the area of fallout and indicate qualitatively the relative intensity of radiation.

Precise determination of the fallout area requires consideration of many complex meteorological parameters. However, from the above analysis a practical field tool can be developed that in most cases will satisfactorily define the area of interest.

Approved by:

E. R. Jomphins

E.R. TOMPKINS
Head, Chemical Technology Division

For the Scientific Director

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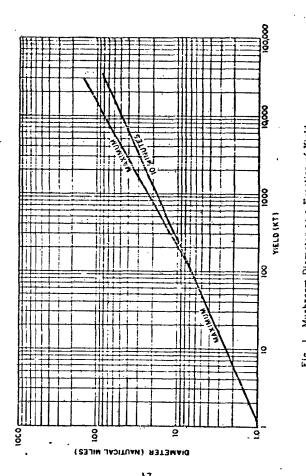
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-15-

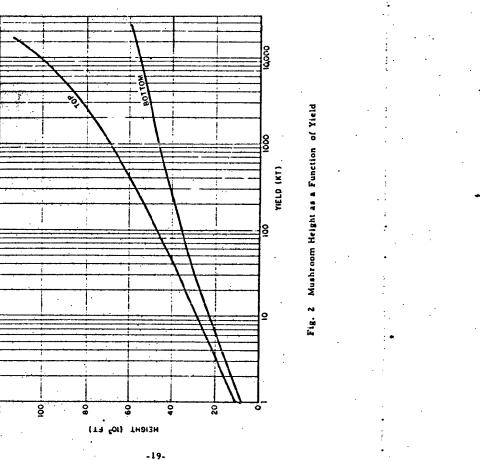
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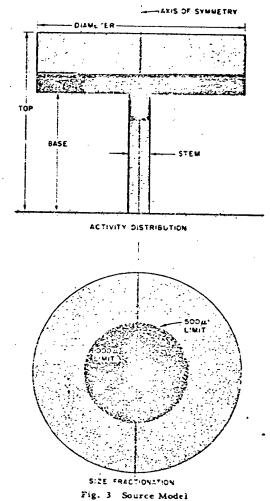
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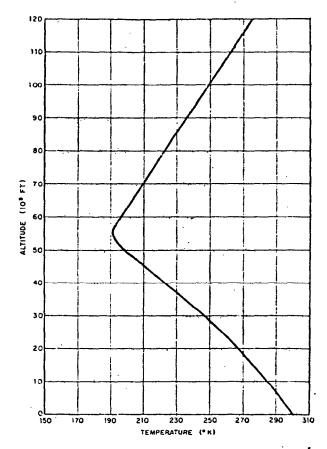


Fig. 4 Temperature as a Function of Altitude for a Marshall Islands Atmosphere

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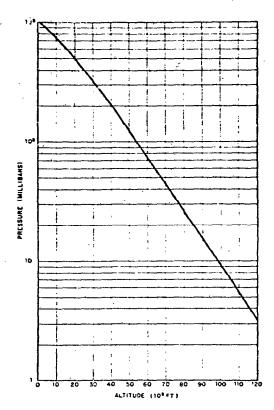


Fig. 5 Pressure as a Function of Altitude for a Marshall Islands Atmosphere

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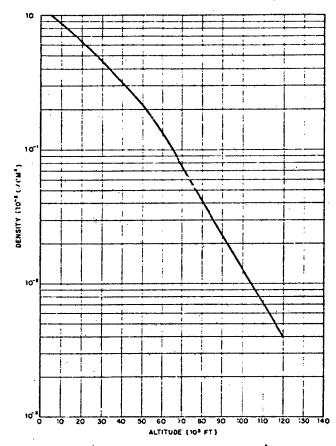


Fig. 6 Density as a Function of Altitude for a Marshall Islands Atmosphere

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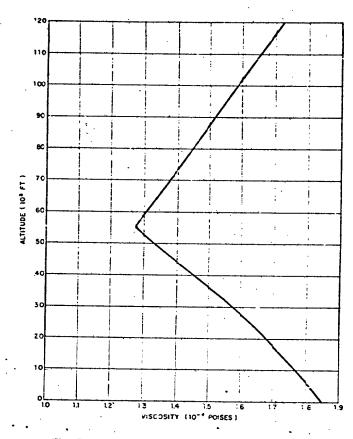


Fig. 7 Absolute Viscosity as a Function of Altitude for a Marshall Islands Atmosphere

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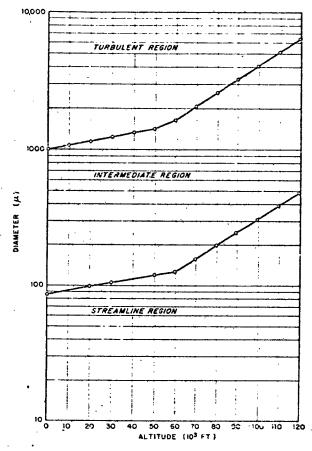
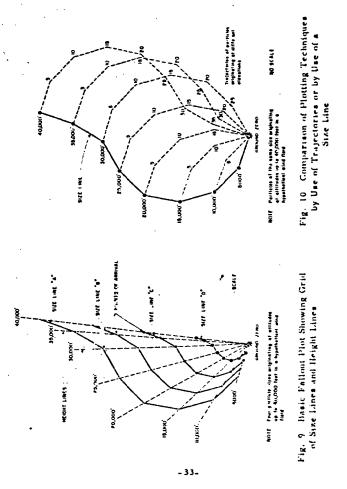


Fig. 8 Falling Speed Transition Zones for the Marshall Islands as a Function of Particle Size and Altitude

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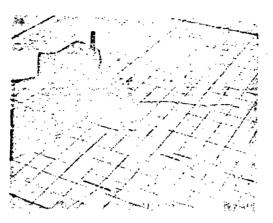
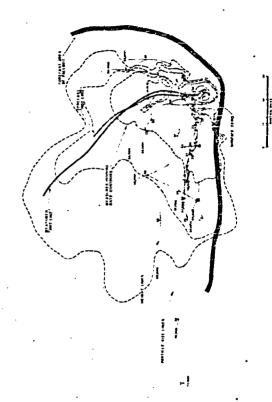


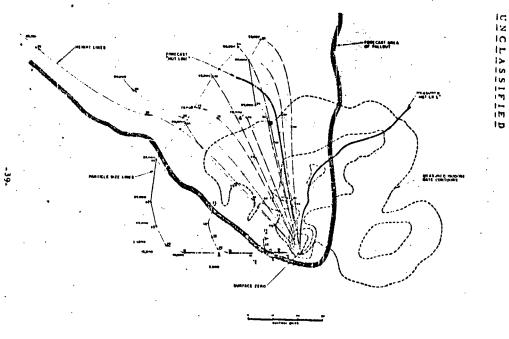
Fig. 11 Fallou: Plotting Device.

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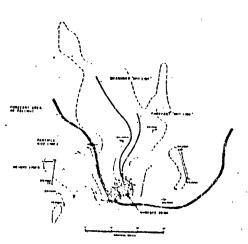


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Comparison of Fallout Forecast With Test Results - Shot B



Comparison of Fallout Forecast With Test Results - Shot C

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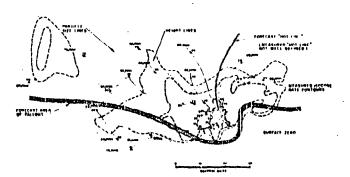


Fig. 15 Comparison of Fallout Forecast With Test Results - Shot D

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TABLE 1

Temperature, Pressure, Density, and Viscosity of the Atmosphere Over the Marshall Islands During the Spring

Altirude (ft)	Temperature (°K)	Pressue (mb)	Density (g cm ³ -10 ³)	Viscosity (poises - 10 ⁴)
SFC .	301	1046	1.17	1_64
1,000	299 -	980	1.13	1.83
2,660	297	950	1.10	1.825
3,050	256	930	1.56	1.815
4,500	295	900	1.03	1.610
5,700	293	870	1.0	1.805
6,500	292	520	5.97	1.795
7,500	290	820	0.24	1.785
8,000	289	90-0	0.91	1,760
9,500	288	776	0.88	1,770
10,066	265	740	9.86	1.765
11,000	284	720	0.83	1,755
12,600	282	€90	9.86	1,745
13,000	280	660	0,78	1.740
14,660	278	640	0.76	1.736
15,000	276	620	0.73	1.726
16,900	274	530	0.71	1.715
17,000	278	576	0.T.6	1.765
18,000	271	550	2.67	1.695
19,565	269	530	9.65	1.655
26,969	267	500	ŭ _63	1,675
21,960	265	450	6,61	1.665
22,000	263	460	0.59	1,655
23,000	261	440	6.57	1.645
24,000	259	426	0.55	1,635
25,000	257	410	0.53	1.625
26,000	255	390	0.52	. 1.615
27,000	252	370	0.50	3,605
28,000	250	35 5	0.49	1,590
29,000	248	340	0.47	2.580
30,000	24€	320	0.45	1.570
31,000	243	310	9,43	1.569
32,000	241	300	0.42	1.545
33,000	239	280	5.42	1.535
34,000	236	270	6.39	1.508
35,000	234	260	0,38	1,510
36,000	232	245	0_37	2,560
37,000	230	235	0_36	1,430

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(Continued)

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TABLE 1 (Continued)

Temperature, Pressure, Density, and Viscosity of the Atmosphere Over the Marshall Islands During the Spring

Alprede (ft)	Tem perature (*Ky	(mb)	(\$_cm10g)	Viscourty (boxes -11 ⁶)
38,666	227	225	1,35	1.478
39,000	225	211	6,53	1,465
40,-300	223	205	·· (,22	: 450
41,000	220	155	(.3)	1,446
42,000	218	185	32.5	1.436
43,000	215	275	. (.23	1,420
44,000	213	285	6,28	1.465
45,500	233	160	1.27	2,225
46,609	263	150	34,7	2,380
47,000	206	145	1.25	3,370
45,600	254	13°	(.54	گ کشر:
49,000	201	130	2ند.)	3,54\$
50,000	189	125	1.22	1,335
51,060	196	115	1.23	1,320
52,060	254	710	. المشا	3.538
53,660	193	105	6,39	1,298
54,000	152	300	1.18	1,285
55,000	451	95	6.17	1,278
56,000	151	96	1.24	1,275
57,000	152	23	1.155	1,280
58,000	193	60	1.145	1.255
59,000	254	77	1.548	1.275
66,000	195	~3	7.135	306
61,003	197	70	(.125	312.:
62,466	198	£ 6 -	1,215	1,320
E3,660	199	£3	111.3	:_205
64,600	201	60	C.168	1,336
65,630	555	56	0.200	1.346
66,000	203	53	6,094	1,345
67,000	205	54	F_C88	2.256
65,606	206	48	1.552	1,360
\$9,000	207	46	C.C7E	1.365
70,660	2.0E	43	£79_3	1,376
71,000	220	41	6-625	1,383
72,000	211	ಚಿತ	6,662	• ودي
73,000	223	37	2.362	1,39.
74,000	214	35	(,હક્ષ્	1.451
75,000	215	33	6,154	1.465

(Cost work)

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TABLE 1 (Continued)

Temperature, Pressure, Density, and Viscosity of the Atmosphere Over the Marshall Islands During the Spring

Altitude (ft)	Temperature (°K)	Pressure (ii.b)	(g'cm ³ , 10 ³)	(poises-104)
76,000	217	32	0.052	1.415
77,000	218	30	2,649	1.420
78,000	219	28	0.546	1.430
79,000	221	27	0.044	1.435
£0,000	222	26	0.642	1.449
81,600	223	24	0.039	1.450
82,000	225	23	0.037	1,455
83,000	226	22	0,434	1.465
84,000	227	21	. 0.032	1.470
85,000	2.79	26	0.030	1,482
36,000	236	19	0.029	1.485
87,000	231	18	0.027	1,490
88,000	233	17	0,026	1,500
89,600	234	16	0.024	1,605
90,000	2?5	15	0.023	1,510
91.000	237	14.5	3,021	1,520
92,000	238	14	2,925	1.525
93,000	239	13	0.019	1.535
94,000	241	12,5	e.c.a	1,540
95,000	242	12	0.017	1.550
96,000	243	11	0.016	1,555
97,000	245	10,5	0,015	1,565
98,000	246	10	0,614	1,570
99,000	247	9.5	9.013	1,575
100,060	249	. 9	0.013	1,585
101,000	250	8.5	8,010	1_590
102,000	251	8	0.010	1,500
103,000	253	7.6	0.010	1.605
104,000	254	~.4	0.010	1.613
105,000	255	0	0.0095	7,620
106,000	25" ·	6.6	0.0090	1.625
107,000	258	6.2	0,0055	1.635
108,000 -	259	6.0	0.0080	1.640
109,000	261	5,6	0.0075	1.650
110,000	262	5.4	0,0070	1,655
111,000	2,63	5.1	0,0068	1,660

(Continued)

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TABLE 1 (Concluded)

Temperature, Pressure, Density, and Viscosity of the Atmosphere Over the Marshall Islands During the Spring

Altitude ·	Temperature (^O K)	Pressure (mb)	(fr.cm.y. 10.y) Derreit	Vaccounty (posses 10 ⁴)
112,000	265	4.3	L.IXIG4	1,675
113,000	265	4,6	0.0066	1,675
114,000 .	267	4.4	0,0056	1.685
115,000	269	4.2	0,5654	1,696
116,000	278	3.9	0.0059	1.760
117,000	272	3.*	5.1348	1.705
118,000	273	3.6	0.0544	2,710
119,000	274		2,3642	1.750
120,000	275	3.2	0.0046	1.725

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TABLE 2
Falling Speeds as a Function of Altitude (ft/hr)

Alifrude		Particle Dis	imeter	
(1011)	75 µ	160 д	200 p	25:0
0	3060	- 5640	11,700	21,600
5	3120	5240	12,300	22,960
10	3200	5480	12,300	24,100
15	3270	5750	13,700	25,500
20	3360	5980	14,400	27,100
25	3470	6160	15,300	28,800
30	3570	6380	16,300	30,860
35	3720	6640	17,500	33,500
40	3870	6910	18,600	35,300
45	4040	7260	19,800	37,800
50	4210	7520	21,460	40,660
55	4420	7860	23,200	44.600
60	4260	7760	24,400	47,200
65	4190	7480	26,100	51,160
70	4110	7320	27,660	55,200
75	4010	7150	28,100	59,70
80	3910	v960	27,800	61,900
85	3800	6770	27,100	67,800
90	3720	6640	26,500	71,300
95	3620	6470	25,800	77,300
100	3550	6340	25,300	80,200
105	3470	6180	24,800	75,800
110 .	3400	6050	24,000	74,200
115	3330	5930	23,700	72,600
120	3260	5860	23,460	71,100

Cumulative Time of Fall for the 75-µ Particles (hr)

				Startle	Starting Sievation z 103 fe	501 2 00						
Airmde	120 - 116	116-110	110-105	106-100	100 115	00-98	80-R5	N6-ED	80-75	16.10	16-10 10-65	65-60
120-116	1.62											
116-110	3,01	1.40										
110-105	4.40	2.2	1.46									
196-100	84.3	4.38	2.87	1.43								
100-38	7.27	5.75	4,26	 E	2.1							
98 - PG	6,63	1,1	5.03	↓ .17	2.75	1.36						
80· 89	9.00	1.44	9.00	09.7	4.08	2.60	£.1					
85- BO	11.26	9.74	8.28	6.80	5.38	3.10	7	1,30				
HO- 75	12.52	11.00	9.51	N.00	6.04	0.29	3,80	200	1,20			
75 - 70	13,76	12.23	10.74	02.0	7.87	6.48	. T.	3.75	2.40	1.23		
30	14.15	13.43	11.0M	10.40	0.07	7.68	6,32	7.2	3,69	1,43	1.20	
65. 60	16.14	14.62	13,13	11 GX	02/01	8,87	1,5	£3	4.83	3.03	2,36	68.T
99 .00	17.30	15.78	0.7 ¥.20	¥.:	11.42	00'01	H. 07	7.	Ž,	4.7	3,58	2,35
55 50	18.40	10.94	15,45	14.00	12,68	ET:18	E.8.4	8.10	7,30	3	Ę	3.61
-03	10.67	18.16	16.41	15,21	13,79	12.40	H,9	 	6.4	7.13	5,02	4.73
40-40	20.03	10.41	17.92	18.47	15.06	13,64	12,30	10,17	0.07	#.	7.18	B.98
40 36	22.20	20,73	10.34	UT.71	16,37	¥.98	13,62	12.1	10,99	E. 73	9,50	7.30
96-30	20.62	22,10	20.61	10.16	1:.74	16.35	14.03	1:1,00	12.30	2.1	78.6	8.67
30- 38	26,04	23,62	22.03	20,58	15.16	17,71	16,41	15,48	13,78	12.62	11,29	10,09
26- 20	26.59	24.08	23,49	#3.C#	21.62	10.23	17,87	16.54	10,24	13.08	12,76	1.36
20- 35	#.B.01	26.40	25.00	23.66	52,53	20,74	10,38	18,03	10,70	10.40	14,26	13,06
٥ ٠	29.55	28.03	72.02	25,09	70,62	22.28	20,02	10.00	18.29	17.03	15.80	14.60
ė	31.13	29,61	28, 12	26.67	25.28	23,86	22.50	21,17	19.87	18,61	17,38	10.18
÷	32.75	31,23	29.74	28.10	20.87	26.48	24.12	21.70	21.49	20.23	19.00	17.80

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TABLE 3 (Continued)

Cumulative Time of Fall for the 75-µ Particles (hr)

Intermediate Altitude	60-55	55-50	50-45	45-40 Sta	tting Elev 40-35	36-20	30-25	25-20	20-15	15-10	10-5	5-0
120-315				···								
115-110											٠.	
110-168				•								
108-100												
100- 95												
95- 90												
90- 85											٠.	
85- 80												
80- 75												
75- 70												
70- 65		•		٠.		•						
65- 6u												
60- 55	1.16										•	
55- 50	2.32	1.16										
50- 45	3.63	2.37	1,21									
46- 43	4.79	3.63	2.47	1.26								
40- 35	6.11	4.95	3,79	2.58	1,32							
35- 30	7,48	6,32	5.16	3.95	2,69	1.37						
30- 25	8.90	7,74	6.58	5,37	4,11	2.79	1.42					
25- 20	10,36	9,20	8.04	6.83	8.87	4.25	2,88	1.46				
20- 15	11.87	10,71	33,9	8,34	7.08	5.76	4.39	2.97	1.51			
18- 10	13.41	12.26	11.09	9.88	8.62	7.30	6,93	4.51	3.05	1.54		
10- 5	14,09	13.83	12.67	11,46	10,20	8,88	7,51	6.00	4,63	3.12	1.58	
4 - 0	16.61	15,45	14,29	13.08	11.82	10.52	9,13	7.71	4.03 6.25	4,74	3,20	1.0

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TABLE 4

Cumulative Time of Fall for the 100-µ Particles (hr)

Intermediate				Stat	ting cleva	tion > 10	3 (1					,
Altitude	120-116	114-110	110-108	105-100	100-0 4	89- 40	VÚ- #6	21- BO	80-78	75-70	70-65	65-60
120-115	0.88		····	· · · · · · · · · · · · · · · · · · ·				-	-			
116-110	1.68	0.83								•		
110-105	2.50	1,65	0.82									
105-160	3,30	2.45	1.62	0.80	•							
10095	4.08	3,23	2.40	1,58	0.78							
95- DO	4.54	3.99	3,16	2,34	1.54	0.76						
90- HB	4.58	4.73	1,90	3.98	2.28	1,50	0,74					
86- HO	5.30	5.46	4,63	3.84	3.61	2,23	1.47	0.73				
80- 75	7.42	0.17	5.44	4.52	3.72	2.94	2,18	1,44	0.71			
76- 70	7.71	6.HG	6,03	5.21	4,41	3.63	2.47	2.14	. 1.40	0,69		
70- 85	8.38	7,53	6.70	5,88	6,04	4.30	3,84	2,80	2.07	1,36	C.97	
CS- FU	9.04	8,19	7.36	6.54	5,71	4.00	4.20	3.46	2,73	2,02	1,33	0.66
60 - 100	0.68	8,83	R.00	7.18	6.38	5.60	4.84	4.10	3,37	2,66	1.97	1,30
55 - 50	10,33	9.48	8,65	7.83	7.03	6.20	8.40	4,75	4.02	3,31	2.62	1,48
50- 45	11.01	10.16	9.33	8.54	7,71	0.03	0.17	6,43	4,70	3,99	3,30	2,63
46 40	11.72	10,87	10,04	9 22	8,42	7.64	4 48	6, 14	5.41	4.70	4.01	3.34
40-35	12.46	11.61	10.78	9,96	P. 16	8.38	:.42	0.88	6.45	5.44	4,75	4.08
35-30	13.24	12.39	11.56	10.74	9,94	9.15	8,40	7,68	6,93	0.22	8.63	4.88
30 25	14.03	13.18	4 ' '15	11,53	10,73	9.95	9,19	8,46	1,72	7,01	20,0	8,68
28 20	14,85	14,00	13,17	12,35	11,55	10,77	10,01	9.27	8,54	7.63	7,14	0.47
20- 16	15,70	14,85	14,02	13,20	12.40	11,62	10,46	10,12	9,39	HD.H	7,99	7,32
16- 10	16,69	15,74	14.91	14.00	13,40	12,51	11,7/	11.01	10,28	9,67	84.8	8.21
10- 6	17,52	10,67	16,84	15.02	14.22	13.44	12.66	11.04	11,21	10,50	9.81	9.14
8- 0	18.49	17.64	16,81	15.99	18.19	14.41	13.60	12,91	12.18	11,47	10.78	10,11

TABLE 4 (Continued)

Cumulative Time of Fall for the 100-µ Particles (hr)

Intermediate Altitule	60-05	55-50	50-45	45-40	\$14(Ung	Elevation 35:30	30. 53 tr	52-30	20-18	15-10	10-8	8-0
120-115			·									
118-110												
110-105												
105-100									•			
100- 95												
95- 90												
90- 85												
86- 80												
80- 75	•											
76- 70												
70- 85												
65- 60 .												
60- 85	0.64											
86 - AO	1.29	0.65										
80- 4h	1.97	1.33	83,0									
45- 40	2.88	2,04	1,39	0.71								
40- 35	3,42	2.78	2.13	1.45	0.74							
35- 30	4.20	3.68	2,91	2.23	1.62	0.78						
30- 25	4.99	4.35	3,70	3,02	2.31	1.67	0.70					
25- 20	6.01	8.17	4,62	3,84	3.13	2.39	1.61	0.82				
20- 15	6.66	6,02	8,37	4.69	3,98	3.24	2.46	1.67	48,0			
18- 10	7.55	8.91	8.26	88.8	4.87	4.13	3,34	£.50	1.74	0.89		
10- 6	8,48	7.84	7.19	6,81	5,50	6,06	4.28	3,49	2.67	1.82	0,93	
6· 0	9.40	8,81	8,16	7,48	6.77	6.03	5.28	4.46	3.64	2.70	1,90	0,07

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TABLE 5
Cumulative Time of Fall for 200-µ Particles (hr)

	Imermediate				Stattle	g Clevatic	ni k 100 (i					
	Altitude	120 114	115-110	110-105	108-100	100-96	116 - PU	90.86	40·40	BO-76	78 - 70	70-65	44-6
	120-115	0,21			1								
	110-110	0,42	0.21										
	140-105	0,62	0,41	0.20									•
	105-100	0,82	0,01	0.40	0.20		•						
	100- 50	1,02	0.81	0.06	0.40	0,20							
	95- 90	1,21	1.00	0.78	0,59	0.39	0,19						
	90- 86	1,40	1,19	0.98	0.78	0,68	45,0	0,19					
	85 - 80	1,68	1,37	1.16	0,44	0,70	0,66	9,37	0,18				
	80- 76	1,76	1,66	1.34	1.14	0.04	0,74	0,65	0.36	0,18			
	76 - 70	1.01	1,73	1.52	1,32	1,12	0.94	0.73	0.54	0,34	0,18		
	ንስ፣ ወኔ	2,13	1.99	1.71	1,81	1,31	1.11	0, 97	0.73	84,0	0.01	0,19	
	00 40	2,33	2. 12	1.91	1,71	14,1	1,31	1, 12	0.93	0,78	0,57	0,39	0,20
	80 88	2.64	2, 33	2.12	1,92	1.72	1.62	1.33	1.14	a,us	0.78	0.60	17,41
	03 - 44	2 76	2,66	2.34	2.14	1.121	1,74	t.aa	1,34	1.1#	1,00	0.83	0.03
	80- 48	3,90	2,70	2.58	2.38	2.18	1,98	1,79	1,60	1,42	1.24	1,00	0, M*
	45- 40	3,26	3.65	2,84	2.64	2,44	2,24	7,05	טי, ו	1.38	1,50	1,32	1,1:
	4 / 36	31,54	3,.13	3.42	2,92	2,72	2.02	2,33	\$,11	1,05	1,78	1,60	1.4
	36 - 30	3.14	3,6.1	3.42	3,22	3,02	2,42	2,63	2,44	2.26	2,08	1,90	1,71
	30 - 25	4,10	ล.รถ	3,74	3,64	3,34	3,14	2.95	2.76	2,0A	2,40	2.22	2,0
	28 - 26	4.50	4.20	4,08	ня, с	3,611	3,48	3,29	1,10	2.02	2.74	2.66	8 3,
	20- 15	4,86	4.65	4.44	4,24	4,04	3,84	7,68	3,40	3.28	1.10	2.02	2.7
	15-110	8,24	6.03	4,82	4,62	4,42	4.22	4,93	3,64	.1,66	3,48	1,30	3.1
	10. 1	4,64	5.43	5,22	4.02	4,82	4.62	4,43	4,24	4.00	3,88		3,8
	8- 0	6.08	8,86	8.44	A.44	6,24	1.04	4.46	4.00	4.48	4,30	4.12	3.9:

TABLE 5 (Continued) Cumulative Time of Fall for 200-µ Particles (hr)

nternjedlata					Starting	Blevation	n 103 ft					
Altitude	60-55	66-80	50-45	45-40	40-35	36-30	30-28	30 - 20	20-15	18-10	10-A	8-0
120-115												
115-110												
110-100												
106-100												
100- 95												
96- 90						•						
PD- 86												
AR- ser												
e ;. 7°,												
78 - 70												
70- 63												
C5- 60												
60- 55	0,21											
65- 50	0.43	0,22										
80 45	9.67	0.40	0.24									
46- 40	0.93	0.72	0.61	0.28								
40- 38	1.21	1,90	0,78	0.54	0,28							
35- 30	1.61	1.30	1,04	0.84	0,58	0,30						
30 25	1.80	1.62	1.40	1.16	0,90	0,62	0.32					
26 - 30	2.17	1.96	1,74	1,50	1,24	0.96	0.66	0,34				
20- 18	2,63	2.32	2,10	1.80	1,00	1.32	1.02	0,70	0.36			
16- 10	2.91	2,70	2,48	8.24	1,48	1.70	1.40	1,08	0,74	0.38		
10- 6	3.31	3,10	2,88	2.64	9.38	2.10	1,80	1.44	1.14	0,78	9.40	
5· 0	3.73	3.52	3,30	3.06	2,80	9,62	9,99	1.00	1.84	1.90	0.89	0.42

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. ~ 1, 5 5 %

TABLE 6 Cumulative Time of Fall for 350-p Particles (hr)

Allinde Spuill A	120-110	110-110	110-109	5terti 105 100	ng flevati Lan-us	Бој и по ОЧ-АЧ	ft pg #8	MA · MO	po- 78	76 - 70	70-45	61-0
120 115	0,07		-	***************************************				14 14.	****	-		
116 110	0,14	0,07										
110 105	0.21	0.14	0.07									
105 100	0,27	0,20	0,14	0,06								
100 95	0.33	0,90	0.10	0,14	(1,440							
80 - 90	0,40	0.34	0,20	0.19	0.13	0.07						
Ha - 144	0.47	0.44	5.33	0,70	0,70	0.14	0.07					
AR AN	0.66	0 4R	0.41	0,34	0,48	0,92	0.15	(1,04				
AD- 70	0.43	0.66	0,40	0.42	0.30	0,30	0.24	0.10	0,08			
76 70	0.72	0.60	U.AN	0.61	0.48	0.39	0,32	0.45	6,17	0.09		
70 86	0,41	0.74	0.67	0,60	0.64	(), 4 R	0.41	0.44	0.76	0.18	a, ep	
an an	0,91	0.84	0.77	0.70	11,64	a,68	0.51	0.44	0,36	11,148	0.49	0.10
60 55	1,02	a ph	0 84	0.81	0.15	0.69	0.62	0.66	0.47	0,30	0.30	0.91
0.0 0.0	1,14	1.07	1,00	0.93	0.87	0.81	0.74	0.67	n,Au	0.51	0.44	0,33
80 - 40	1,27	1.70	1,13	1,08	1 00	0.94	0.87	6,40	0,79	11,64	0.50	0,46
17 40	1,41	1.14	1,27	1.20	1.44	6.08	1.01	0,94	0.86	0.78	1.01	0,84
10- 36	(,64	IAN	1,42	1,:10	1,24	1 23	1,10	1,09	1.01	11,93	13,84	0.78
an an	1.79	106	1,64	1,51	1.48	1,39	1.32	1,20	1.17	1.09	1,00	0,01
30 AV	ua, j	1.82	1.76	I.AH	1,42	1.66	1,44	1,49	1.54	1.98	1,17	1.08
26 20	2 07	y pn	1,91	1 84	1,80	1,74	1,47	1.40	I BY	1 44	1,35	1,26
20 16	u 20	2.19	2,12	2.05	1,44	1.95	1,84	1.19	1.71	1,03	1.84	1.40
26 IU	¥, 46	2,39	2.12	2,25	N. 19	8,11	3.06	1.44	1.91	1.43	1,14	1,65
10 A	2.67	2.60	11.0.1	¥.48	2.10	9.34	H.47	2.50	9.18	¥.04	1.00	1,86
n. 0	8.45	2.82	2 70	N.05	2.62	g.ad	2,42	2,42	g.34	2.174	8,17	1.0A

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TABLE 6 (Continued)

Cumulative Time of Fall for 350-p Particles (hr)

	•	line-mediate					Statiling	fievstion	n 105 fc						
		Ailliude	60· Ab	68-50	80-48	45 40	40-16	26 - 30	30 25	37 30	20-15	15-10	10 &	8.0	
		12 - 113				~									
		115 110					•								
		110-168													
		106 100													
		100 95													
		PG - 44													
		AN .OG													
		08 48													
	:	## 7A													
	Š.	75 70													
	•	70- \$6											:		
	•	02 40													
•		ፅ ፀ - ልል	0.11												
		bh · Au	0.23	0,12											
		80 - 45	0.34	0.24	0,13										
		46- 40	0,50	0.34	0,27	0.14									
		40 - 35	0,44	0,04	0,42	0,20	0,10								
		38. 30	0,81	0,70	9,68	0.44	0.31	v, 1d							
		48 06	0,98	0,87	0,78	9.52	U.48	EE,0	0 17						
		89+ 80	1,18	t.oh	C, PH	0,80	0.44	0,61	0.35	0.1#					
		20- 16	1.16	1,94	1,12	n,up .	0,85	0.70	0.64	9,17	0, 10				
		18- 10	1,00	1.44	1,38	1.10	1,06	0,90	0,74	0,07	0,39	0.40	•		
		10. 0	1.76	1,65	1.63	1,40	1.86	1.11	0,04	0. 18	Q,du	0.41	18.0		
		á· 0	1,08	1,87	1.75	1.42	1,48	1,33	1,17	1,00	0,92	0.62	6,43	0, 22	
						• • •		*****		•					
		A SHELD IN THE PROPERTY.						AND DESCRIPTION OF	* *1 (** - ****** **	-		* * ***			

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